

NUMERICAL SIMULATION OF SUPERSONIC MIXING IN A BURROWS–KURKOV COMBUSTOR BY USING SA-RANS METHOD

R. S. Solomatin^{1,2} and I. V. Semenov^{1,2}

¹ Scientific Research Institute for System Analysis, Russian Academy of Sciences, 36-1 Nakhimovskiy Pr., Moscow 117218, Russian Federation

² Institute for Computer Aided Design, Russian Academy of Sciences, 19/18 Brestkaya 2nd Str., Moscow 123056, Russian Federation

Abstract: Numerical model of mixing of parallel spatial turbulent supersonic flows is developed. The Reynolds-averaged Navier–Stokes (RANS) approach, based on Spalart–Allmaras (SA) turbulent model supplemented with a turbulent diffusion mixing model is used. The averaged Navier–Stokes equations system closed with the turbulence model equation is solved with the LU–SGS–GMRES (lower-upper symmetric Gauss–Seidel generalized minimal residual) method. For the SA turbulence model and turbulent diffusion model, numerical algorithms validation in multicomponent gas mixtures, modeling of hydrogen injection into the $M = 2.44$ inert gas flow, and their further mixing in the model Burrows–Kurkov combustor is conducted. The problem is solved in two- and three-dimensional cases. The results are compared with available experimental and computational data. The calculations are performed on “MVS-10P” JSCC RAS cluster.

Keywords: supersonic flows; mixing; Spalart–Allmaras turbulence model; Burrows–Kurkov combustor; LU–SGS–GMRES algorithm

DOI: 10.30826/CE19120308

Acknowledgments

This work was supported by the subsidy given to the Scientific Research Institute for System Analysis to implement the state assignment on the topic No. 0065-2019-0005 “Mathematical modeling of dynamic processes in deformable and reactive media using multiprocessor computational systems” (Registration No. AAAA-A19-119011590092-6).

References

1. Frolov, S. M., V. S. Aksenov, V. S. Ivanov, S. N. Medvedev, and I. O. Shamshin. 2018. Flow structure in rotating detonation engine with separate supply of fuel and oxidizer: Experiment and CFD. *Detonation control for propulsion: Pulse detonation and rotating detonation engines*. Eds. J.-M. Li, C. J. Teo Boo Cheong Khoo, J.-P. Wang, and C. Wang. Springer International Publishing AG. 39–59.
2. Smetanyuk, V. A., S. M. Frolov, V. S. Ivanov, and B. Basara. 2019. Vliyaniye sposoba podachi toplivnykh komponentov na kharakteristiki detonatsionnogo raketnogo dvigatelya [The influence of the method of supplying fuel components on the characteristics of a detonation rocket engine]. *Goren. Vzryv (Mosk.) — Combustion and Explosion* 12(2):74–84.
3. Solomatin, R. S., I. V. Semenov, and I. S. Menshov. 2018. Towards calculating turbulent flows with the Spalart–Allmaras model using the LU–SGS–GMRES algorithm. Keldysh Institute for Applied Mathematics preprints 119. 30 p.
4. Burrows, M. C., and A. P. Kurkov. 1973. Analytical and experimental study of supersonic combustion of hydrogen in a vitiated airstream. NASA-TM-X-2828.
5. Edwards, J. R., J. A. Boles, and R. A. Baurle. 2012. Large-eddy/ Reynolds-averaged Navier–Stokes simulation of a supersonic reacting wall jet. *Combust. Flame* 159:1127–1138.
6. Spalart, P. R., and S. R. Allmaras. 1992. A one-equation turbulence model for aerodynamic flows. AIAA Paper No. 1992-0439. 22 p.
7. Turbulence Modelling Resource of Langley Research Center. Available at: <https://turbmodels.larc.nasa.gov> (accessed August 29, 2019).
8. Burcat, A., and B. Ruscic. 2005. Third Millennium Ideal Gas and Condensed Phase Thermochemical Database for Combustion with updates from Active Thermochemical Tables. ANL-05/20 and TAE 960 Technion-IIT. Aerospace Engineering and Argonne National Laboratory, Chemistry Division.
9. Menshov, I. S., and Y. Nakamura. 2004. Hybrid explicit-implicit, unconditionally stable scheme for unsteady compressible flows. *AIAA J.* 42(3):551–559.
10. Luo, H., J. D. Baum, and R. Lohner. 2001. An accurate, fast, matrix-free implicit method for computing unsteady flows on unstructured grids. *Comput. Fluids* 30:137–159.

11. Godunov, S. K., A. V. Zabrodin, M. Ya. Ivanov, *et al.* 1976. *Chislennoe reshenie mnogomernykh zadach gazovoy dinamiki* [Numerical solution of multidimensional problems of gasdynamics]. Moscow: Nauka. 400 p.
12. Van Leer, B. 1979. Towards the ultimate conservative difference scheme. V — A second-order sequel to Godunov's method. *J. Comput. Phys.* 32:101–136. doi: 10.1016/0021-9991(79)90145-1.s
13. Gao, Z., Ch. Jiang, and Ch.-H. Lee. 2016. On the laminar finite rate model and flamelet model for supersonic turbulent combustion flows. *Int. J. Hydrogen Energ.*, 41:13238–13253.

Received August 27, 2019

Contributors

Solomatin Roman S. (b. 1994) — junior research scientist, Institute for Computer Aided Design, Russian Academy of Sciences, 19/18 Brestkaya 2nd Str., Moscow 123056, Russian Federation; Scientific Research Institute for System Analysis, Russian Academy of Sciences, 36-1 Nakhimovskiy Pr., Moscow 117218, Russian Federation; roman.solomatin.94@gmail.com

Semenov Ilya V. (b. 1973) — Candidate of Science in physics and mathematics, leading research scientist, Institute for Computer Aided Design, Russian Academy of Sciences, 19/18 Brestkaya 2nd Str., Moscow 123056, Russian Federation; leading research scientist, Scientific Research Institute for System Analysis, Russian Academy of Sciences, 36-1 Nakhimovskiy Pr., Moscow 117218, Russian Federation; semenov@icad.org.ru